National Transportation Safety Board Washington, DC 20594

Brief of Accident

Adopted 08/26/2003

AD01LA048

File No. 13777	04/26/2	001 Plattsburgh, NY	Aircraft Reg No.	Aircraft Reg No. N974FE		Time (Local): 19:45 EDT	
Engine Ma Aircraft Number o Operating Cer Name Type of Flight (t Damage: Subst f Engines: 1 rtificate(s): On-de of Carrier: Wiggi Operation: Non-s	Canada / PT6114 antial emand Air Taxi	Crew Pass	Fatal 0 0	Serious 0 0	Minor/None 1 0	
Ď	Last Depart. Point: Same as Accident/Incident Location Destination: Albany, NY Airport Proximity: Off Airport/Airstrip			Condition of Light: Night/Bright Weather Info Src: Weather Observation Facility Basic Weather: Visual Conditions Lowest Ceiling: None Visibility: 10.00 SM Wind Dir/Speed: 170 / 006 Kts Temperature (°C): 9 Obstr to Vision: None Precipitation: None			
Pilot-in-Command	Age: 46			Flight Ti	me (Hours)		
Certificate(s)/Rating(s) Airline Transport; Multi-engine Land; Single-engine Land; Single-engine Sea Instrument Ratings Airplane			-	Total All Aircraft: 9144 Last 90 Days: 137 Total Make/Model: 137 Total Instrument Time: 2841			

The pilot said the preflight, engine start, run-up, taxi and takeoff were "normal". The pilot said that during the climb after takeoff, approximately 1,000 to 1,500 feet above the ground, the airplane's engine "spooled down, slowly and smoothly, like a loss of torque or the propeller going to feather." The pilot performed a forced landing to a field, where the airplane nosed over, and came to rest inverted. Examination of the engine and propeller revealed that the propeller-reversing lever was installed on the wrong side of the reversing lever guide pin, and that the reversing linkage carbon block was no longer installed, and had departed the airplane. Examination of the airplane's maintenance records revealed that the carbon block was replaced during a 100-hour maintenance inspection, 5 hours prior to the accident. Installation of the reversing lever on the incorrect side of the guide pin resulted in improper seating and premature wear of the carbon block. According to the engine manufacturer, any disconnection in operation of the propeller control linkage will cause the propeller governor beta control valve to extend, and drive the propeller into feather.

Brief of Accident (Continued)

IAD01LA048

File No. 13777 04/26/2001 Plattsburgh, NY Aircraft Reg No. N974FE Time (Local): 19:45 EDT

Occurrence #1: LOSS OF ENGINE POWER Phase of Operation: CLIMB - TO CRUISE

Findings

1. (C) PROPELLER GOVERNOR CONTROL, LINKAGE - DISCONNECTED

2. (C) MAINTENANCE, INSTALLATION - INCORRECT - COMPANY MAINTENANCE PERSONNEL

3. (C) PROPELLER FEATHERING - UNCONTROLLED - PILOT IN COMMAND

Occurrence #2: FORCED LANDING
Phase of Operation: DESCENT - EMERGENCY

Occurrence #3: IN FLIGHT COLLISION WITH TERRAIN/WATER

Phase of Operation: EMERGENCY DESCENT/LANDING

Findings

4. TERRAIN CONDITION - OPEN FIELD

Findings Legend: (C) = Cause, (F) = Factor

The National Transportation Safety Board determines the probable cause(s) of this accident as follows.

The incorrect installation of the propeller reversing lever and carbon block assembly, which resulted in a loss of propeller thrust.